



Multi-disciplinary Design Optimization (MDO) of Airborne Radome with Evolutionary Algorithms

Dr MR Shankar, CABS, DRDO



Topics

- Introduction
- Problem statement
- MDO architecture and analyses
- Results
- Discussion
- Future





Introduction

- Centre for Airborne Systems (CABS) is under DRDO, Ministry of Defence
- Airborne Surveillance Systems and Flight Test Research









- All these systems are multidisciplinary in nature
 - Aerodynamics, Structures, Electromagnetics,... integrated approach
 - MDA & O is very much essential



Introduction

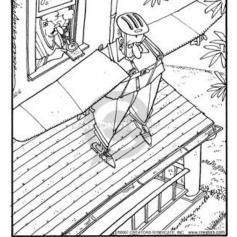


* All figures are illustrative

Design point: Aircraft flying at 0.4

Problem statement

- To optimise airborne radome with respect to aerodynamic, structural and electromagnetic design aspects using Multidisciplinary Design Optimization (MDO) principles
- Process
 - Modeling disciplinary aspects of radome for numerical simulation
 - Integrate disciplinary analyses on a software framework capturing the inter-disciplinary interactions
 - Carry out Multidisciplinary Design Optimization (MDO) involving three disciplines i.e., aerodynamics, structure and electromagnetics
 - Study the trend of disciplinary objectives with respect to each other
 - Establish Pareto optimality for three objectives



'Thumbs up, Arthur, our glide calculations still check out!"



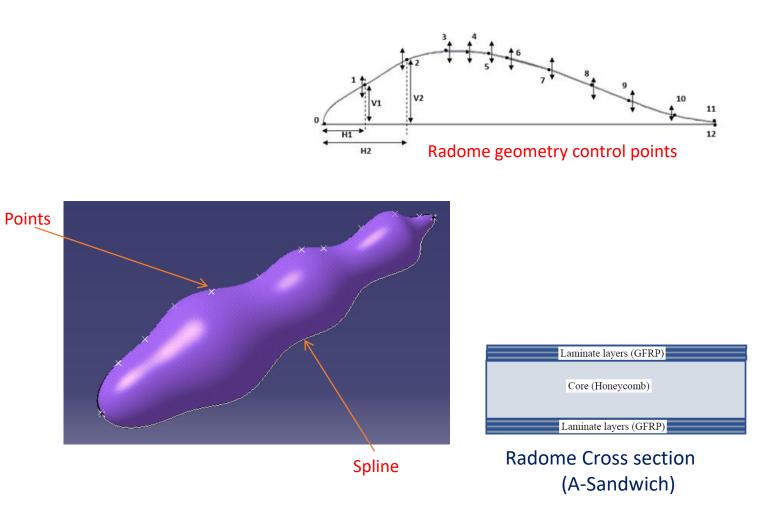
Problem statement

DESIGN VARIABLES

Points 1 to 10: x and y coordinates (20) Point 11 : y value (1) Point 12 : x value (also radome length)

Layers in outer skin (1) Layers in inner skin (1) Thickness of core (1)

Total = 25



Problem statement

DESIGN CONSTRAINTS

Strength constraint (Tsai Wu index)

$$F_1\sigma_1 + F_2\sigma_2 + 2F_{12}\sigma_1\sigma_2 + F_{11}\sigma_1^2 + F_{22}\sigma_2^2 + F_6\tau_{12} + F_{66}\sigma_6^2 < 1.0$$

Antenna size : 0.5m x 0.25m

Geometry constraint

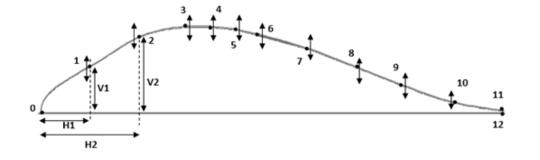
For distance 0.5m to 1.0m (H_i) from leading edge; the height of radome (V_i) should be less than or equal to 0.25m (to accommodate antenna)

DESIGN OBJECTIVES

Drag force (J₁) $D = \frac{1}{2}C_D\rho v^2 S$

Radome weight (J_2)

$$W = SA_R * \left[(N_L * T_L * \rho_L) + (T_c * \rho_c) \right]$$

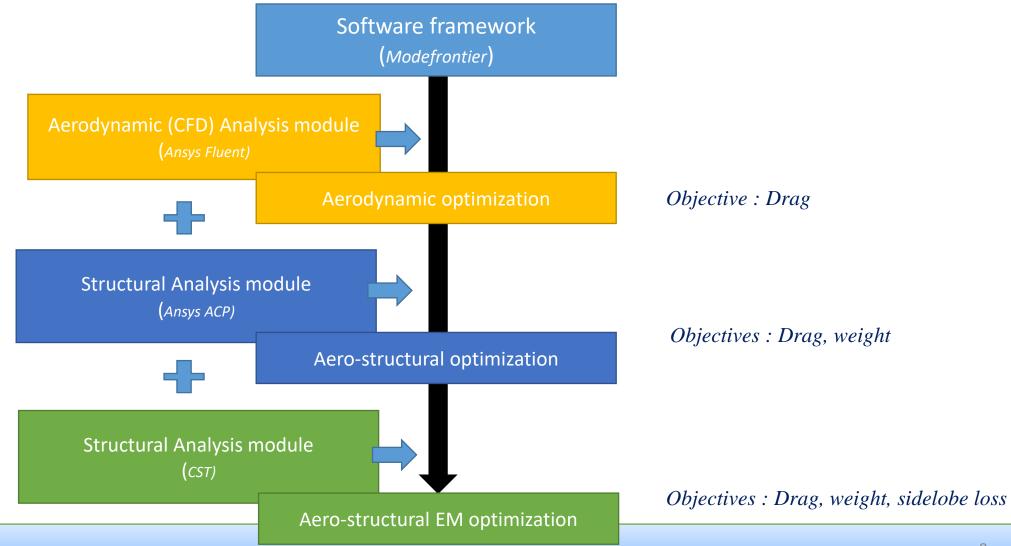


Sidelobe loss (J₃)

E = (Sidelobe level without radome - Sidelobe level with radome)



MDO architecture and analyses



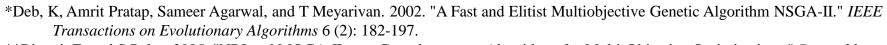


MDO architecture and analyses

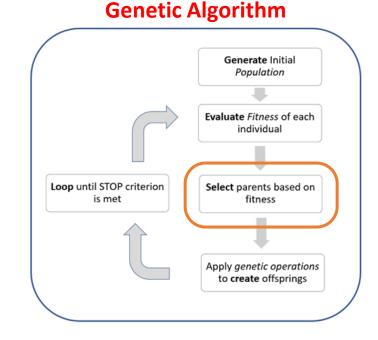
- GA is the chosen evolutionary algorithm

Three algorithms used

- Non Sorting Genetic Algorithm II (NSGA II)*
- Deb et al (2002)
- Multi Objective Genetic Algorithm II (MOGA II)**
 - Rigoni and Poles (2005)
- PilOPT
 - Proprietary algorithm *modefrontier*

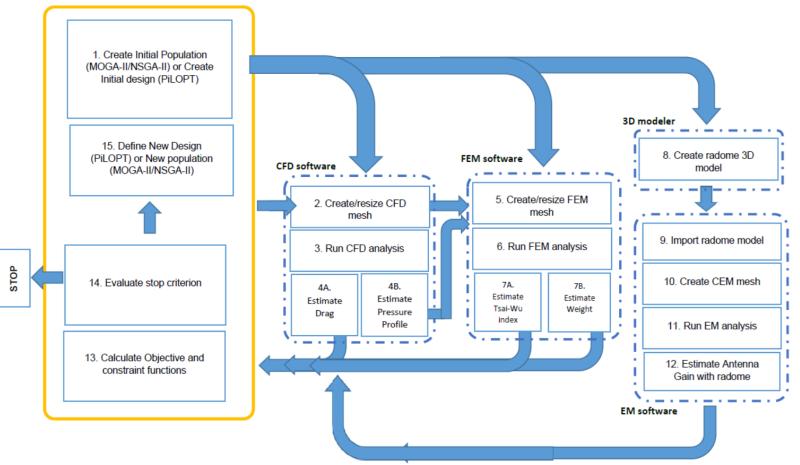


**Rigoni, E, and S Poles. 2005. "NBI and MOGA-II, two Complementary Algorithms for Multi-Objective Optimizations." *Dagstuhl* Seminar Proceedings

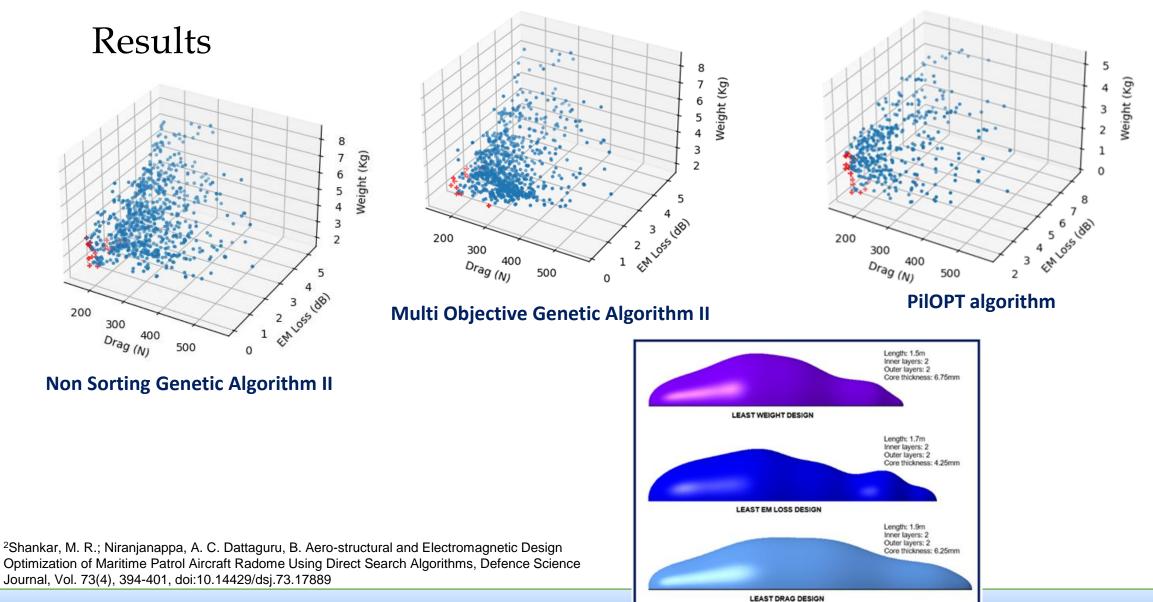


MDO architecture and analyses

Optimization software









Discussions

- Initially behavior of aerodynamic and structural disciplines studied
 - Are in the expected line
 - Well defined 2D Pareto front established¹
 - *Modefrontier* enables read-in / read out of communication and results thru well established routines
- Three disciplines of radome design² integrated and *MDO* carried out.
 - Arguably for the first time for radome.
 - CAD model is read thru CST and EM model created; thru routines from *Modefrontier*
- 3D Pareto optimality explored and achieved for first time
 - Large number of designs for decision maker to choose from
- Fully automated MDO process for radome design and optimization
 - Can be used for other external structures of aircraft as well

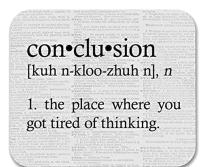
²Shankar, M. R.; Niranjanappa, A. C. Dattaguru, B. Aero-structural and Electromagnetic Design Optimization of Maritime Patrol Aircraft Radome Using Direct Search Algorithms, Defence Science Journal, Vol. 73(4), 394-401, doi:10.14429/dsj.73.17889

¹Shankar, M. R.; Niranjanappa, A. C. & Dattaguru, B. Aerodynamic and Structural Optimization of Maritime Patrol Radar System Radome using Evolutionary Algorithms, Defence Science Journal, 71(4), 2021, 421-428, <u>doi:10.14429/dsj.71.16249</u>



Conclusions

- Two well-used GA techniques applied for MDO; both give similar results.
 - NSGA is able to find marginally more designs on *Pareto front* than MOGA
- *Modefrontier* proprietary code, PilOPT, also yields same results.
- Fully automated MDO process for radome design and optimization
 - Can be used for other external structures of aircraft as well





Future work

- These are starting points for an industrial solution
 - Deeper analyses of chosen designs (in Pareto front) required
- To include manufacturing and environmental aspects
 - Manufacturing tolerances affecting design output
 - Coatings and Paints play an important role in EM; they need to be modelled as well